

# **Boundary Layer Separation Control Using Vortex Generator Jets**

(ジェットによる境界層の能動制御)

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## はしがき

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### 研究費

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### 研究発表

#### (1) 学会誌等

- ・ 長谷川裕晃、松内一雄、田中淳介：ジェット吹出しの縦渦発生装置による能動的はく離防止フィードバックシステムの開発、ターボ機械、26巻12号、1998、746-753.
- ・ Hasegawa, H., Matsuuchi, K., and Yamakami, J.: The Mechanism of Active Boundary Layer Control Using Vortex Generator Jets, ICAS-98-3.4.3.
- ・ 長谷川裕晃、松内一雄：ジェット吹出しによる縦渦発生装置の吹出し角の違いがはく離防止効果に与える影響、日本機械学会論文集（B編）、64巻627号、1998、3690-3696.
- ・ Hasegawa, H., Matsuuchi, K., and Tanaka, J.: Development of an Active Separation Control System Using Vortex Generator Jets, Proc. 3<sup>rd</sup> ASME/JASME Joint Fluids Eng. Conf., FEDSM99-6944, 1999
- ・ 長谷川裕晃、松内一雄、田中淳介：パルスジェット吹出しによる縦渦の生成メカニズム、日本機械学会論文集（B編）、巻号未定、2000

#### (2) 口頭発表

- ・ 田中淳介、長谷川裕晃、松内一雄：縦渦発生ジェットを用いた能動的はく離防止システムの開発、日本機械学会流体工学部門講演会講演論文集、No.98-15、1998、53-54.

- ・ 田中淳介、長谷川裕晃、松内一雄：パルス噴出しの縦渦発生ジェットが縦渦に及ぼす影響、日本機械学会北陸信越支部第36期講演会講演論文集、No.997-1、1999、349-350.
- ・ 田中淳介、長谷川裕晃、松内一雄、小松崎泰寛：縦渦発生ジェットによる境界層内の渦構造、日本機械学会東海支部第49期総会講演会講演論文集、No.0001-3、2000、203-204.

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